

NONLOCAL DAMAGE MODEL USING THE MESHLESS FINITE POINTS METHOD

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Abstract. *We present the formulation and application for a nonlocal model with gradient type regularization incorporated into the model by means of non-local displacements.*

The strong formulation of the FPM allows the use of high-order differentially shape functions with which we can approximate directly the fields of the nonlocal displacements, for that this technique is very attractive for a computational viewpoint.

For the numerical implementation we used a fully explicit integration scheme and for the nonlinear problem the Newton Raphson iterative scheme. The validation of the obtained results is made starting from typical benchmark problems and available results on associated literature.

Keywords: *Non-locas damage models. Meshless. gradient models.*

1. INTRODUCTION

The finite points method (FPM) was proposed by Oñate et al [5, 6] initially with the purpose of solving convective transport and fluid flow problems. Later, its application was extended to advection-diffusion transport [7] and incompressible flow problems [8]. In the context of solid mechanics, FPM has been applied successfully in elasticity [9, 20, 10, 21], solid dynamics [11] and non-linear material behavior problems [13]. The non-dependence on a mesh or integration procedures is an important aspect which transform the FPM in a truly meshless method.

The continuous damage models may be employed to describe the evolution of failure processes between the undamaged state and macroscopic crack initiation [23]. Regarding numerical simulations, this presents pathological mesh sensitivity, in that way different solutions have been proposed in the literature to remedy this physically unrealistic behavior by means cohesive crack models [24], Crack bands model [25] and regularized models[26]. In this work we focus on the regularized models via non-local effects, incorporated a material

characteristic length and the implicit gradient-enhanced continuum model based on non-local displacements [27].

For the gradient-enhanced model the strong formulation of the FPM allows the use of high-order differentially shape functions with which we can approximate directly the fields of local and non-local displacements [28].

This paper contains: Section 2, The basic formulation of the FPM is presented. Section 3 The nonlocal damage model via nonlocal displacements. Section 4 describe the numerical implementation. Section 5 show the classic numerical examples, this examples are in 1D and 2D.

2. The finite points method

Although the FPM introduced by Oñate et al. was originally formulated for the numerical solution of convective transport and fluid flow problems, it can be easily adapted to the Stress equation. In this section, we review the basic formulation of the FPM and provide a brief overview of its main features. In order to obtain the final system of discrete equations, the FPM approximates the local solution of a partial differential equation in each point of the discretized domain by means of a weighted least squares technique and a point collocation procedure. Due to the local character of the approximation procedure used by this method, it is necessary to define a sub-domain for each node, Ω_k , that contain neighboring nodes selected by a suitable criterion [18, 9]. This collection of points is called a *cloud*, and its referential central point is the *star node*. For example, a relevant aspect in the definition of *clouds* is that their superposition must produce the whole domain, Ω .

$$\bigcup_{k=1}^{N_p} \Omega_k = \Omega , \quad (1)$$

where N_p is the total number of nodes. Note that the definition of *clouds* is the basic, initial step in implementing the FPM approximation using fixed weighted least squares. With the discretized domain defined, let us define a function $u(x)$, which is locally approximated by $\hat{u}(x)$ only valid in the *cloud* Ω_k associated with the *star node* x_k as a linear combination of known functions $\mathbf{p}(x)$.

$$u(x) \cong \hat{u}(x) = \mathbf{p}(x)\boldsymbol{\alpha}_k \quad \forall x \in \Omega_k , \quad (2)$$

where $\mathbf{p}(x)$ is the vector that represents the basis of m linearly independent functions, and $\boldsymbol{\alpha}_k$ is a vector of constant parameters only valid in Ω_k . The elements of the interpolation base may belong to any function family. Nevertheless, for simplicity, the m first monomial polynomials are used. Since Eq. (2) is valid for all N_c points of the k -th sub-domain, the approximations $\hat{\mathbf{u}}(\mathbf{X})$ conform to a *Vandermonde* system given by the following relation.

$$\mathbf{u}(\mathbf{X}^k) \cong \hat{\mathbf{u}}(\mathbf{X}^k) = \mathbf{P}(\mathbf{X}^k) \cdot \boldsymbol{\alpha}_k , \quad (3)$$

where

$$\begin{aligned} \mathbf{X}^k &= [x_{k,1} \dots x_{k,N_c}]^T, & \mathbf{u}(\mathbf{X}^k) &= [u(x_{k,1}) \dots u(x_{k,N_c})]^T, \\ \hat{\mathbf{u}}(\mathbf{X}^k) &= [\hat{u}(x_{k,1}) \dots \hat{u}(x_{k,N_c})]^T, & \boldsymbol{\alpha}_k &= [\alpha_{k,1} \dots \alpha_{k,N_c}]^T, \end{aligned}$$

and

$$P(\mathbf{X}^k) = \begin{bmatrix} \mathbf{p}(x_{k,1}) \\ \vdots \\ \mathbf{p}(x_{k,N_c}) \end{bmatrix}.$$

In general, the number of points N_c that conform to the *cloud* is greater than the number of functions m that define the basis; hence, the matrix $P(\mathbf{X}^k)$ is usually rectangular. This means that the property of interpolation is lost, and the problem must be addressed with numerical approximation. The coefficients of the vector α_k must be determined in such a way that the weighted sums of the squared differences between the exact values $u(x)$ and the approximated values $\hat{u}(x)$ of each point is minimized according to the following equation.

$$\min \left\{ \sum_{j=1}^{N_c} w(x_j) \cdot (u(x_j) - \hat{u}(x_j))^2 \right\}, \quad (4)$$

where $w(x_j)$ is a fixed weighting function defined in Ω_k and evaluated for the node x_j . See refs. [5, 6]. The minimization process described by Eq. (4) leads to the following expression for vector α_k .

$$\alpha_k = A^{-1}(\mathbf{X}^k) B(\mathbf{X}^k) \lambda(\mathbf{X}^k), \quad (5)$$

where $\lambda(\mathbf{X}^k)$ is a vector that represents the unknown parameters sought on the *cloud* Ω_k defined as follows.

$$\lambda(\mathbf{X}^k) = [\lambda(x_{k,1}) \dots \lambda(x_{k,N_c})]^T, \quad (6)$$

Additionally, matrix $A(\mathbf{X}^k)$, $B(\mathbf{X}^k)$ and $W(\mathbf{X}^k)$ are given as follows.

$$A(\mathbf{X}^k) = P(\mathbf{X}^k) W(\mathbf{X}^k) P^T(\mathbf{X}^k), \quad B(\mathbf{X}^k) = P^T(\mathbf{X}^k) W(\mathbf{X}^k),$$

and $W(\mathbf{X}^k)$ is an $N_c \times N_c$ diagonal matrix defined by:

$$W(\mathbf{X}^k) = [\text{diag}(w(x_{k,1}) \dots w(x_{k,N_c}))], \quad (7)$$

where the weighting functions $w(x_j)$ are derived in order to have unit values near the *star node* and zero values outside the Ω_k sub-domains. Under the FPM, the common selection is the normalized *Gaussian* given as follows.

$$w(x_j) = \begin{cases} \frac{\exp(-(h_j/\zeta)) - \exp(-(r/\zeta))}{(1 - \exp(-(r/\zeta)))} & \text{if } h_j \leq r \\ 0 & \text{if } h_j > r \end{cases}, \quad (8)$$

where h_j is the distance between the *star node*, the point x_j , $r = q \cdot h_{max}$ (max. of h_j) is a reference distance, and $\zeta = \beta \cdot r$. A detailed description of the effects of the constant parameters q and β on numerical approximation as well as guidelines for setting their values are presented in [19]. Other considerations in selecting the function $w(x_j)$ can be found in refs. [6, 5, 20, 22]. Finally, replacing Eq. (5) in (2), the next relation is obtained.

$$\hat{u}(x) = N^T(x) \lambda(\mathbf{X}^k), \quad (9)$$

where $N(x)$ is a matrix called *shape function* defined by:

$$N(x) = \mathbf{p}^T(x) \mathbf{C}(\mathbf{X}^k), \quad (10)$$

with $\mathbf{C}(\mathbf{X}^k) = \mathbf{A}^{-1}(\mathbf{X}^k) \mathbf{B}(\mathbf{X}^k)$. Note that according to the least square nature of the approximation, $u(x) \cong \hat{u}(x) \neq \lambda(x)$. That is, the local values of the approximating function do not fit the nodal unknown values. Indeed, $\hat{u}(x)$ is the true approximation, which we will use to satisfy the differential equation and the boundary conditions; in this context, $\lambda(x)$ are simply the unknown parameters we aim to determine. According to the concepts described above and Eq. (9), it is possible to obtain the following expressions.

$$\hat{u}_x(x) = N_x^T(x) \boldsymbol{\lambda}(\mathbf{X}^k) \quad \text{and} \quad \hat{u}_{xx}(x) = N_{xx}^T(x) \boldsymbol{\lambda}(\mathbf{X}^k), \quad (11)$$

where $(\cdot)_x$ and $(\cdot)_{xx}$ denote the first and the second space derivatives, respectively. Note that these derivatives are computed by taking the derivative of the basis functions $\mathbf{p}(x)$ in Eq. (2).

3. The nonlocal damage model

The so called continuum damage model have been used thoroughly to simulate the behavior of materials that present degradation of the mechanical properties due to small fissures that appears during the loading process. To characterize this, the concept of effective stress $\bar{\sigma}$ is introduced. In one dimension we can write

$$\sigma = (1 - d) \bar{\sigma} \quad (12)$$

Where $\bar{\sigma}$ is the effective stress and d is the damage parameter between $0 \leq d \leq 1$. The stress with the effective deformation ε can be related by Hooke's law:

$$\bar{\sigma} = E \varepsilon \quad (13)$$

Where E is the elastic modulus of the material, then replace (12) in (13) we have:

$$\sigma = (1 - d) E \varepsilon \quad 0 \leq d \leq 1 \quad (14)$$

for the nonlocal damage model based on non-local displacements [26], equation 14 takes the following form

$$\sigma(\varepsilon_a, \varepsilon_g) = (1 - d(\varepsilon_g)) C : \varepsilon_a \quad (15)$$

where $\varepsilon_g = \nabla u_g$ is the gradient-enriched strain and $\varepsilon_a = \nabla u_a$ is the strain associated to the local displacement field u_a . also the displacement field u_g is the solution of the diffusion-reaction equation (16) in which u_a is the source term

$$u_g - l^2 \nabla^2 u_g = u_a \quad (16)$$

with the following boundary conditions

$$u_g \cdot \mathbf{n} = u_a \cdot \mathbf{n} \quad \text{on } \Gamma_u \quad (17)$$

$$\nabla(u_g \cdot \mathbf{t}) \cdot \mathbf{n} = \nabla(u_a \cdot \mathbf{t}) \cdot \mathbf{n} \quad \text{on } \Gamma_t \quad (18)$$

the damage parameter $d(\varepsilon_g)$, depends on the non-local strain ε_g via the history variable κ , defined as

$$\kappa(t) = \max_{\tau \leq t} Y(t) \quad (19)$$

where Y is a scalar state variable. In one dimensional setting, Y is simply the scalar strain ε_g . In a multi-dimensional setting, Y is defined either as

$$Y = \sqrt{\sum_i \max(0, \varepsilon_i)} \quad (20)$$

where ε_i are the principal strains of ε_g or as

$$Y = \frac{k-1}{2k(1-2v)} I_1 + \frac{1}{2k} \sqrt{\left(\frac{k-1}{1-2v} I_1\right)^2 + \frac{12k}{(1+v)^2} J_2} \quad (21)$$

where I_1 and J_2 are the first invariant and the second deviatoric invariant respectively, v the Poisson's ratio and k the ratio of compressive to tensile strength.

In this work we use a linear evolution law for the damage parameter

$$d(\kappa) = \begin{cases} 0 & \text{if } \kappa \leq k_i \\ \frac{k_u(\kappa - k_i)}{\kappa(k_u - k_i)} & \text{if } k_i < \kappa \leq k_u \\ 1 & \text{if } \kappa > k_u \end{cases} \quad (22)$$

where k_i is the damage initiation strain and k_u is the ultimate strain. This model is suitable for quasi-brittle materials.

4. Numerical implementation

Considerer the system of differential equations which governs the behavior of a solid

$$\nabla \sigma(\mathbf{x}) + \rho \mathbf{b}(\mathbf{x}) = 0 \quad \forall \mathbf{x} \in \Omega \quad (23)$$

$$\sigma(\mathbf{x}) \cdot \hat{\mathbf{n}} = \bar{\mathbf{t}}(\mathbf{x}) \quad \forall \mathbf{x} \in \Gamma_t \quad (24)$$

$$\mathbf{u}(\mathbf{x}) = \hat{\mathbf{u}}(\mathbf{x}) \quad \forall \mathbf{x} \in \Gamma_u \quad (25)$$

using the point collocation method, we can obtain a discrete system of equations [20]

$$\mathbf{L}^T \sigma + \rho \mathbf{b} = 0 \quad \forall \mathbf{x} \in \Omega \quad (26)$$

$$\mathbf{n}^T \sigma = \bar{\mathbf{t}} \quad \forall \mathbf{x} \in \Gamma_t \quad (27)$$

$$\mathbf{u} = \hat{\mathbf{u}} \quad \forall \mathbf{x} \in \Gamma_u \quad (28)$$

where \mathbf{L} is an operator that defines the differential equation, \mathbf{n} contains the normal direction on the external contour, σ is the stress vector, \mathbf{u} is the displacement vector, $\rho \mathbf{b}$ the body forces, $\bar{\mathbf{t}}$ and $\hat{\mathbf{u}}$ prescribed force and displacement on the external contour.

To obtain an equivalent system in term of displacement, use the stress-strain relation and the strain-displacement relationships as follows [29].

$$\sigma = (\mathbf{I} - \mathbf{d}(\varepsilon_g)) \mathbf{C} \varepsilon_{\mathbf{a}} \quad \text{en donde} \quad \varepsilon_{\mathbf{a}} = \mathbf{N}_{\mathbf{x}} \mathbf{u}_{\mathbf{a}} \quad (29)$$

$$\left[(\mathbf{I} - \mathbf{d})\mathbf{L}^T \mathbf{CLN} + (\mathbf{I} - \mathbf{d})(\mathbf{LN})^T \mathbf{CLN} - (\mathbf{LN}\mathbf{d})^T \mathbf{CLN} \right] \mathbf{u}^h = -\rho \mathbf{b} \forall \mathbf{x} \in \Omega \quad (30a)$$

$$[\mathbf{n}^T \mathbf{CLN}] \mathbf{u}^h = \bar{\mathbf{t}} \quad \forall \mathbf{x} \in \Gamma_t \quad (30b)$$

$$[\mathbf{N}] \mathbf{u}^h = \bar{\mathbf{u}} \quad \forall \mathbf{x} \in \Gamma_u \quad (30c)$$

with its compact form

$$\mathbf{K}_{dI} \mathbf{u}^h = \mathbf{f}_I \quad I = 1, \dots, N \quad (31)$$

Where \mathbf{K}_d is the stiffness matrix in FPM and \mathbf{f} contain the equilibrium and boundary conditions.

4.1. Approximation of diffusion-reaction equation

consider the numerical implementation obtained in (31) and the diffusion-reaction equation (16), to approximate the diffusion term, we used the second-order shape function directly in the same way as in the lineal elastic behavior, then we can write the following stiffness matrix equation for the non-local problem

$$\mathbf{K}_{NL I} = \begin{pmatrix} \mathbf{K}_{dI} & \mathbf{0} \\ -\mathbf{N} & (\mathbf{N} - l_c^2 \mathbf{D}) \end{pmatrix}$$

where \mathbf{D} is the diffusivity matrix defined as

$$\mathbf{D} = \sum_{k=1}^i \frac{\partial^2(\cdot)}{\partial k^2} = \sum_{k=1}^i N_{kk}(\cdot) \quad (32)$$

and the compact form

$$\mathbf{K}_{NL I} \mathbf{u}^h = \mathbf{f}_I \quad (33)$$

In this case $\mathbf{u}^h = (\mathbf{u}_a^h \mathbf{u}_g^h)^T$ and \mathbf{f} contain the conditions for the equilibrium and the diffusion-reaction equation.

5. Numerical examples

For the examples to develop we considered a regular points discretization. The number of points in the interpolation clouds will be specific to each problem.

Table 1. Uniaxial tensile test

Description	Symbol	Value
Length of the bar	L_0	100
Length of weaker part	l_i	10
Cross-section of bar	A	1
Youn's modulus	E	20000
Damage threshold	ϵ_0	10^{-4}
Final strain	ϵ_f	$1.25 \cdot 10^{-2}$

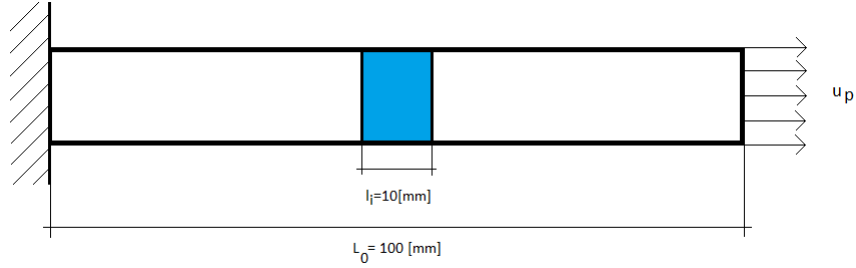


Figure 1. One dimensional rod model

5.1. Uniaxial tensile test

The example consists of a rod subjected to uniaxial tensile load by displacement control as show fig. 1. The central tenth of the bar is weakened 10% by a reduction in Young's modulus to force the localization . The dimensionless geometrical and mechanical parameters are summarized in Table1.

In the first analysis a fixed internal length $l_c = \sqrt{5}$ and five different discretization are used, in a second analysis a fixed number of nodes and four different internal length are used. The results in fig. 2 indicates that discretization dependency is removed and fig. 3 show the effects of the internal length size. These results are equivalent to those reported by [26].

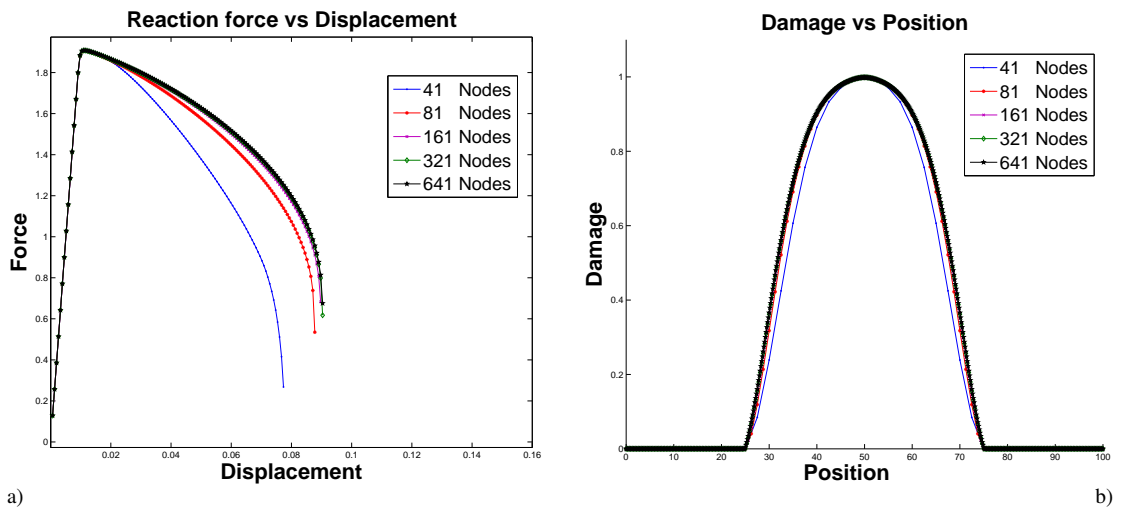


Figure 2. Influence of discretization size on uniaxial tensile test: a) force-displacement curve . b) damage profile .

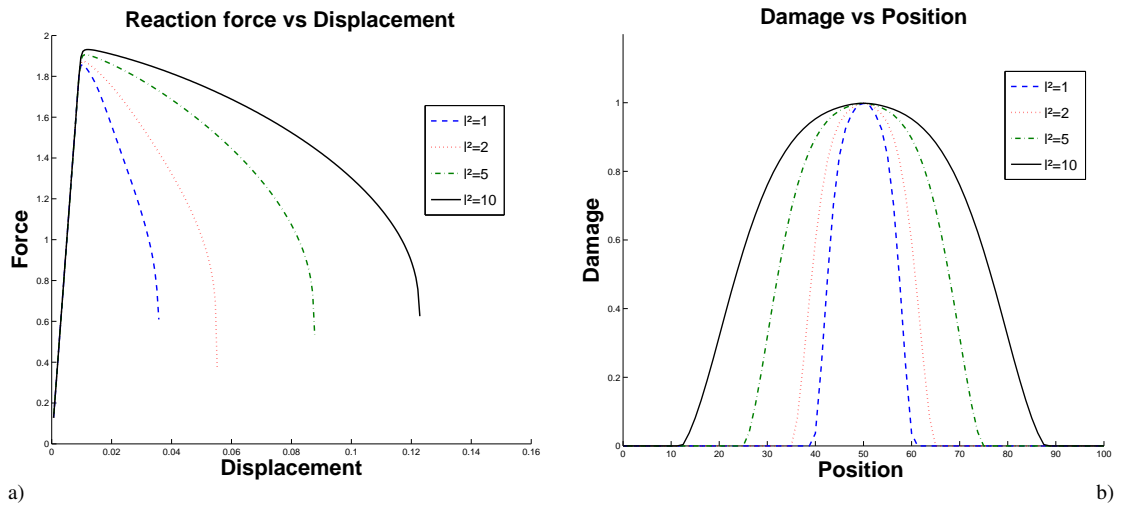


Figure 3. Influence of the internal length size on uniaxial tensile test: a) force-displacement curve . b) damage profile .

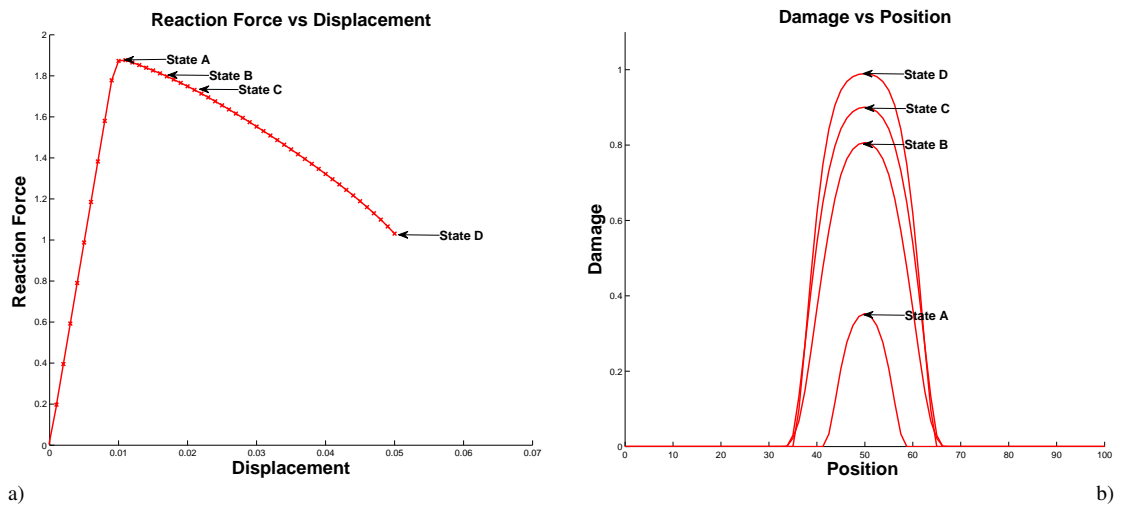


Figure 4. Evolution of the reaction force and internal damage parameter: a) force-displacement curve . b) damage profile .

The final analysis is made with 81 nodes and a internal lenght $l = \sqrt{2}$. The goal is show the evolution of the reaction force, the internal damage parameter and the stress-strain curves for different points.

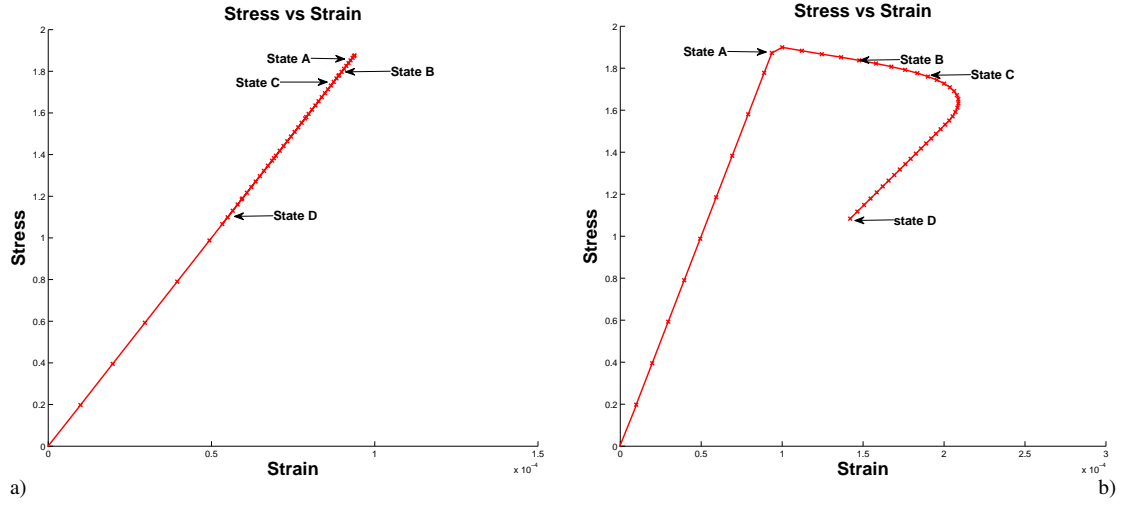


Figure 5. Evolution of the Stress vs Strain at sample points and internal damage parameter: a) $x = 0$. b) $x = 40$.

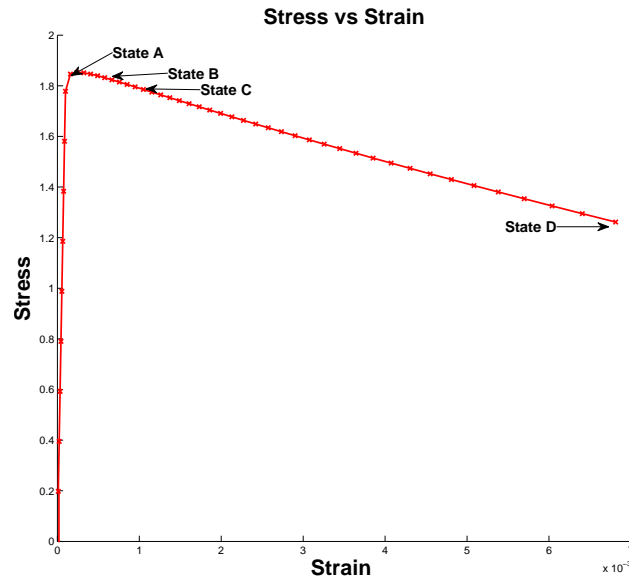


Figure 6. Evolution of the Stress vs Strain at sample points and internal damage parameter at $x = 50$

5.2. Elastic damage tensile specimen with symmetric imperfection

A rectangular specimen is subjected to uniform displacements at the two ends. In order to trigger localization from a homogeneous state of deformation, symmetric material imperfection with weakened material properties are introduced as show in fig. 7, fig. 8 show the results for different discretizations, these results are equivalent to those reported by [30].

The dimensionless material parameters are $E = 2e6$ $\nu = 0.18$ $k_i = 10^{-4}$ $k_u = 0.00625$ and $l = \sqrt{4.5}$.

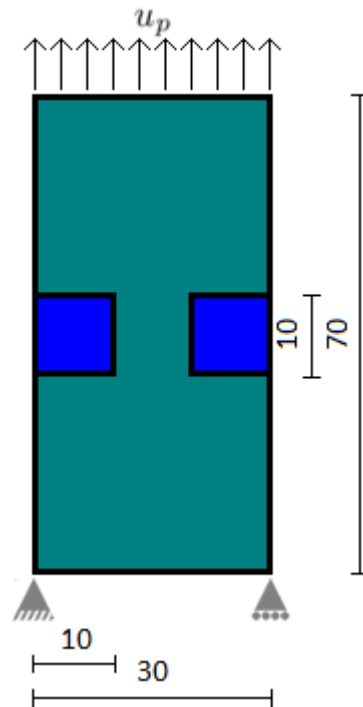


Figure 7. Elastic damage test specimen with symmetric imperfection

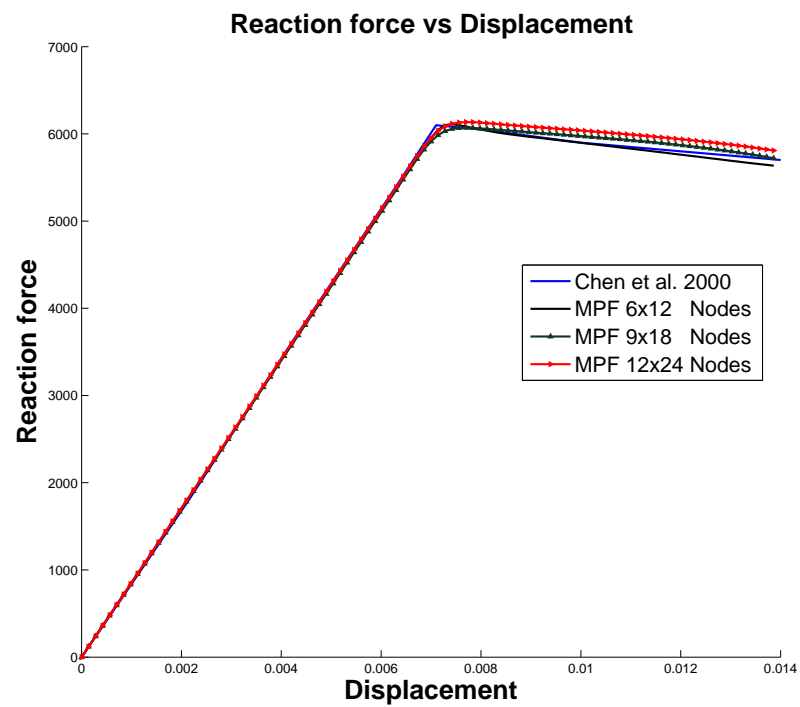


Figure 8. Reaction force vs displacement for various discretizations.

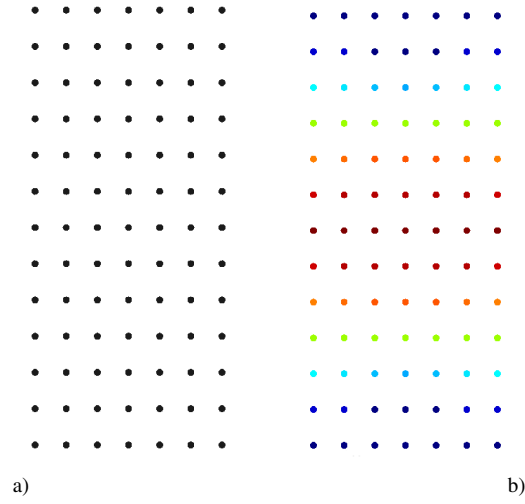


Figure 9. a) analysis model 6x12 nodes . b) strain localization using MPF.

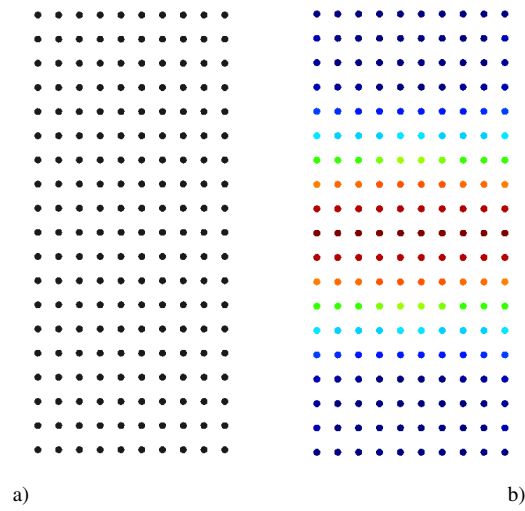


Figure 10. a) analysis model 9x18 nodes. b) strain localization using MPF.

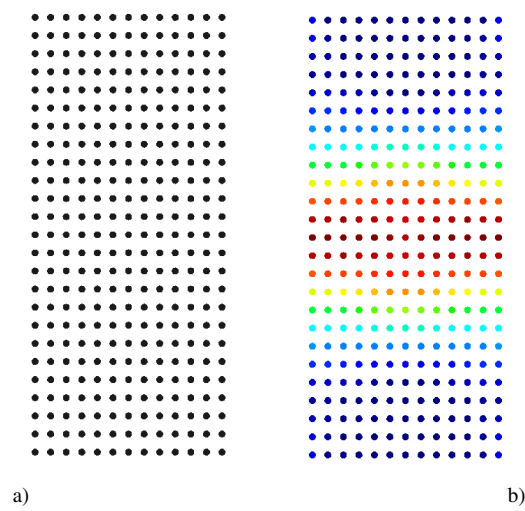


Figure 11. a) analysis model 12x24 nodes . b) strain localization using MPF.

5.3. Elastic damage tensile specimen with asymmetric imperfection

The principal goal in this example is the study of the shear band formation, an asymmetric imperfection is introduced as show in fig. 12. In this only a refined model is used to examine the proposed method. The material parameters are the same as those used in example 5.2, except $\nu = 0.3$ and the internal length $l = \sqrt{0.01}$.

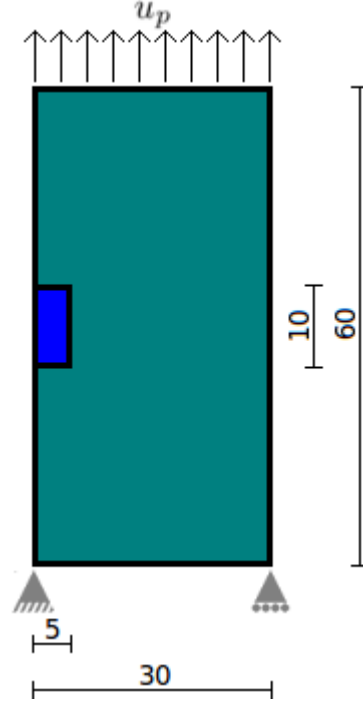


Figure 12. Elastic damage test specimen with asymmetric imperfection

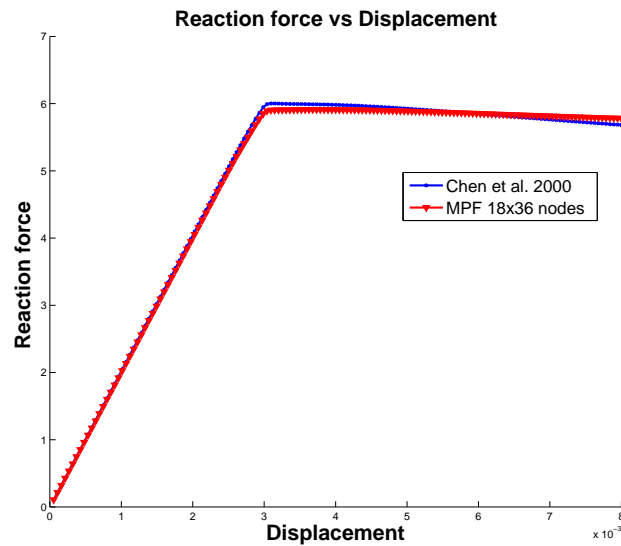


Figure 13. Reaction force vs displacement

Figure 13 show the result with the proposed method, this results is similar to the reported by [30] and fig. 14 shows the shear band formation.

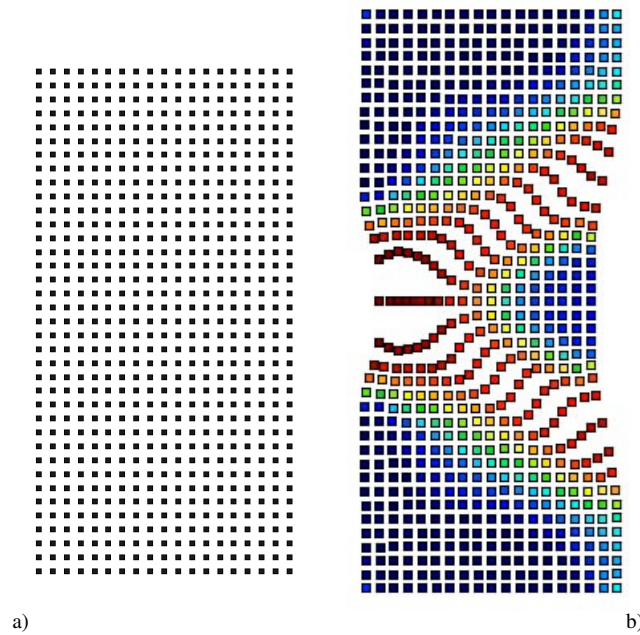


Figure 14. a) analysis model 18x36 nodes . b) strain localization using MPF. deformation scale x100.

6. Conclusion

Has been implemented a isotropic damage model based on nonlocal displacements in strong form by the meshless finite points method to simulate the non-linear material behavior. As the results demonstrate the FPM is able to approximate the localization phenomena with fracture energy regularization by the consideration of characteristic material length and the diffusion-reaction equation. This works concludes the opening of a new alternative viewpoint to the classical weak formulation based on MEF. For a computational aspect, the use of the shape function to approximate all the fields and the simplicity of the development algorithm can be highly attractive.

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